

UNIVERSITI PUTRA MALAYSIA

THE EFFECT OF THERMO-OXIDATIVE AGING ON PROPERTIES OF EGLASS FIBER-REINFORCED EPOXY COMPOSITES

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THE EFFECT OF THERMO-OXIDATIVE AGING ON PROPERTIES OF E-GLASS FIBER-REINFORCED EPOXY COMPOSITES

By

AMIN KHAJEH

Thesis Submitted to the School of Graduate Studies, Universiti Putra Malaysia, in Fulfillment of the Requirements for the Degree of Master of Science

May 2016

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DEDICATION

To my beloved parents

For their

Love

Endless support

&

Encouragement

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Abstract of thesis presented to the Senate of Universiti Putra Malaysia in fulfillment of the requirement for the Degree of Master of Science

THE EFFECT OF THERMO-OXIDATIVE AGING ON PROPERTIES OF E-GLASS FIBER-REINFORCED EPOXY COMPOSITES

By

AMIN KHAJEH

May 2016

Chairman : Faizal Bin Mustapha, PhD, PEng Faculty : Engineering

The present study aims to investigate the effect of thermo-oxidative aging on the mechanical, chemical, physical properties of EHG250-68-37 E-glass fiber-reinforced epoxy preimpregnated.

To achieve the proposed research objectives, laminates of EHG-68-37 fiberglass/epoxy prepreg were exposed 800 h in isothermal condition and aircirculating oven at 82 °C. It is noteworthy that, before aging, specimens were dried to constant weight under vacuum at 70 °C in accordance with ASTM D 5229/D 5229M due to hydrophilic matrix. The variations of mechanical properties (the elastic moduli, tensile strength, strain break, and toughness) were quantified by conducting tensile tests on both aged and un-aged specimens based on ASTM D3039. Chemical changes in composites due to thermo-oxidative aging were analyzed by, Dynamic mechanical analysis (DMA), Differential Scanning Calorimetry (DSC), and Fourier Transform Infrared spectroscopy (FTIR). Physical degradation mechanisms resulting from sub-T_g aging were monitored by weight loss measurements as a function of time and Scanning Electron Microscope (SEM) to investigate superficial resin, cross sectional, and the cryofractured surface morphology.

The showed the toughness, tensile strength and modulus of the composites were increased after pronounced aging conditions, 3.7%, 48%, and 59%, respectively. Whereas a decrease (0.22%) was observed in the strain break. DMA results revealed that the glass transition temperature and rubbery state modulus were increased as a result of the matrix densification. FTIR spectroscopy demonstrated the formation of carbonyl compounds around IR band 1735 cm⁻¹ due to oxidation of the chemical structure of the aromatic ethers. SEM observations indicated the existence of minor superficial cracking, growth in size and number of voids, and poor fiber-matrix adhesion after aging. In addition, a minor mass change was observed from mass loss monitoring methods. The overall findings suggest that post-curing and oxidation enhanced the brittleness of the resin, leading to a significant decline in the useful structural life of the thin-skinned composite.

Abstrak tesis yang dikemukakan kepada Senat Universiti Putra Malaysia sebagai memenuhi keperluan untuk Ijazah Master Sains

KESAN PENUAAN TERMO-OKSIDATIF TERHADAP SIFAT-SIFAT KOMPOSIT EPOKSI DIPERKUKUH GENTIAN KACA-E

Oleh

AMIN Khajeh

Mei 2016

Pengerusi : Faizal Bin Mustapha, PhD, PEng Fakulti : Kejuruteraan

Kajian ini bertujuan untuk mengkaji kesan penuaan termal-oksidatif pada sifat mekanikal, kimia, fizikal EHG250-68-37 gentian kaca-E bertetulang pracampuran epoksi. Untuk mencapai objektif kajian yang dicadangkan, lamina daripada EHG-68-37 gentian kaca / pracampuran epoksi didedahkan dalam keadaan isoterma dan ketuhar pada suhu 82°C selama 800jam. Perlu diperhatikan bahawa, sebelum penuaan, spesimen telah dikeringkan pada satu jisim kekal dibawah vakum pada suhu 70°C mengikut ASTM D 5229 / D 5229M kerana matriks bersifat hidrofilik. Kepelbagaian sifat mekanik (modulus elastik, kekuatan tegangan, titik putus terikan, dan keliatan) telah diukur dengan menjalankan ujian tegangan di keduadua spesimen yang melalui proses penuaan dan tidak melalui proses penuaan berdasarkan ASTM D3039. Perubahan kimia dalam komposit disebabkan oleh penuaan termal-oksidatif dianalisis mengunakan Analisis Mekanikal Dinamik (DMA), Differential Scanning Calorimetry (DSC), dan Jelmaan Fourier Spektroskopi Inframerah (FTIR). Mekanisme kemusnahan fizikal yang disebabkan oleh Sub-Tg penuaan dipantau berdasarkan pengurangan berat terhadap masa dan Mikroskop Imbasan Elektron (SEM) untuk menganalisis keadaan permukaan resin, keratan rentas, dan permukaan morfologi cryofractured. Itu menunjukkan kekuatan, kekuatan tegangan dan modulus bagi komposit meningkat, masing-masing sebanyak 3.7%, 48%, dan 59%. Manakala penurunan (0.22%) diperhatikan pada titik putus terikan. Keputusan DMA mendedahkan bahawa suhu peralihan kaca dan modulus elastik telah meningkat akibat pemadatan matriks. FTIR spektroskop pembentukan sebatian karbonil sekitar band IR 1735cm⁻¹ kerana pengoksidaan struktur kimia eter aromatik. Pemerhatian SEM menunjukkan adanya keretakan kecil pada permukaan, pertumbuhan saiz dan bilangan lubang, dan sedikit gentianmatriks selepas penuaan. Di samping itu, melalui kaedah pemantauan kehilangan jisim, jisim mengalami sedikit perubahan. Hasil kajian menunjukkan bahawa dan pengoksidaan telah meningkatkan kerapuhan resin, yang pengawetan membawa kepada penurunan yang ketara dalam kitaran penggunaan struktur komposit nipis



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LIST OF ABBREVIATIONS

G_I^C	Critical energy release rate/ fracture toughness at Mode I
O_2	Oxygen molecule
P.	Free radical molecule
РОО [.]	Peroxide radical
1,3-BAC	1,3-bisaminomethylcyclohexane
ASTM	American Society for Testing and Materials
BMI	Bismalemide
CFRP	Carbon fiber reinforced plastic
CGMCs	Carbon and graphic matrix composites
CIM	Confocal interferometric microscopy
CMCs	Ceramic-matrix composites
CNC	Computer Numerical Control
C _p	Heat capacity in constant pressure
DGEBA	Diglycidyl ether of bisphenol-A
DMA	Dynamic Mechanical Analysis
DSC	Differential scanning calorimetry
E'	Storage modulus
E"	Loss modulus
EDX	Energy-dispersive X-ray
EPN	Epoxy Phenol Novolac
FTIR	Fourier transform infrared spectroscopy
GFRPs	Glass fiber reinforced plastics
ITX	Isopropylthioxanthone
KBr	Potassium Bromide
LMPA	Low molecular polyamide
MMCs	Metal matrix composites
MTHPA	Methyl tetra hydro phthalic anhydride
OMCs	Organic-matrix composites
PAA	Polyamideoamine
PEEK	Poly-ether-ether-ketone
PEI	Poly –ether-imide
PEKK	Poly-ether-ketone
PMCs	Polymer matrix composites
POPA	Polyoxyproylene
SEM	Scanning electron microscopy
SiC	Silicon Carbide
Tan δ	Mechanical damping; Tan $\delta = E'''/E'$
TEPA	Tetraethylenepentamine
T _g	Glass transition temperature
TMA	Thermal mechanical analysis

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CHAPTER 1

INTRODUCTION

In the chapter, the background of the study, statement of the problem, research objectives, research scope and significance of the research study to the researcher and engineering community are covered.

1.1 Research background

There has been sustained interest on PMCs (Polymer Matrix Composites) from a wide range of specialists especially in recent decades. Initially, the application of composites in the military sparked their commercial usage that began after World War II. New landscape pressures couple with the energy crisis experienced in the 1970s prompted military and commercial aircraft manufacturers to employ the lightweight composite structures. Furthermore, the advancements achieved in the composite area have permitted considerable decrements of weight in the structural design. In comparison to the metal alloys, composites are superior in regard to stiffness and strength to weight ratio, corrosion resistance and outstanding fatigue properties.

Even though the exhibit desirable mechanical properties and low density, concerns arise over the overall and long-term durability that polymer matrix composites have, especially in relation to the load bearing to sustain performance under increased temperatures. Generally, elevated temperatures are likely to cause reversible changes, that is, physical aging, and the non-reversible changes, that is, chemical aging, to be experienced in PMCs. Particularly, combining elevated temperatures and oxygen-induced non-reversible changes is likely to result in considerable decrease in properties. Such non-reversible variances are commonly called 'thermo-oxidative aging' and oxygen diffusion into the composite initiates them.

Numerous research works have been revealed that thermal oxidative aging may change the chemical structure of polymer matrix composites. Alterations to the chemical structure during thermo-oxidative degradation include post-curing [1-11], loss of volatiles [12-17], dehydration [4, 18, 19], chain scission [15, 20-24], additional crosslinking [2, 4], and carbonyl growth [8, 19]. The initial chemical changes are accompanied by dehydration of secondary alcohols and the release low molecular weight gaseous species due to random chain scission. However, at moderate heat exposures (24° C - 177^{\circ}C), chain crosslinking is the dominant chemical change in the matrix compared to chain scission [25]. Indeed, the increase in the crosslink density of the cured matrix primarily occurs during the initial aging period and are caused by post-cure reactions [4, 6, 26, 27], which result in the excessive brittleness in the matrix [28]. As aging proceeds in the presence of oxygen, susceptible chemical structures in the resin are oxidized to various carbonyl containing groups. Therefore, matrix embrittlement increases

with the oxygen concentration and aging time [12, 29]. Extensive research efforts have revealed that the matrix embrittlement enhance bulk mechanical properties such as tensile strength and modulus [30-32] and toughness [33, 34]. However, some studies observed a reduction in toughness resulted from the brittleness in the matrix [35, 36]. According to Colin et al. [37], changes observed in the bulk characteristics of the polymer matrix are caused by the superficial oxidized layer forming as a result of liberated segments volatile being oxidized in the thermo-oxidative aging phase [38-41]. On establishing this, numerous comprehensive research efforts were undertaken on polymer matrix material for investigating the brittle attributes that the matrix exhibit through the thickness. The findings of these studies showed that the matrix embrittlement is confined to the superficial layers and the core of the matrix remained intact [13, 38, 42, 43].

1.2 Problem Statement

Oxidation takes place only in the superficial layers of composites due to their exposure to aging while core of the aged composites mostly remain intact. Considering the advent in applying thin-skinned composite of approximately 1-6 mm in thickness in aeronautical structure, it becomes necessary to ask whether relying on specimen thickness in characterizing bulk attributes in evaluating durability would be an issue. Considering this, Tsotsis [44] remarked that using specimens that are too thick will delay the onset of observable changes in many properties because may be sufficient unchanged material to carry mechanical loads, such that loss or reduction in properties of a material's outer layers masks the degradation. Cinquin and Medda [45] studied the influence of laminate thickness on composite durability for long term. They conducted a durability evaluation on carbon/epoxy laminate with 5.15 mm and 26 mm thick over period of 30,000 h and at temperature of 150°C. The findings of the study indicated that the residual mechanical properties are more affected on thin composite than on thick composites.

In the context of "durability", there is an aspect of both physic-chemical and mechanical effects determining the life time of the considered composite materials [46]. Thus, it is necessary that the thin-skinned composites be characterized on the constitutional level to be able to understand their chemical, mechanical and physical responses caused by thermo-oxidative processes. Nevertheless, the investigation hereby undertaken to evaluate the durability of the thin-skinned composites focused on measuring weight loss and mechanical characterization. Furthermore, even though the usage of the woven fiberglass/epoxy prepreg composites that are thin-skinned has continued rising, the thin-skinned composite remain under-researched when it comes to its durability when subjected to thermo-oxidative aging scenario. Therefore, the primary aim of conducting this thesis was to examine how thermo-oxidative aging affect the durability of the woven fiberglass/epoxy prepreg composites with a thin skin in relation to the physical, chemical and mechanical aspects and in establishing the origin of in-service failure that arise prematurely.

1.3 Research Objectives

This study mainly seeks to examine the influence that thermo-oxidative aging has on the durability of the EHG250-68-37 woven fiberglass/epoxy prepreg as well as the resultant effect on the mechanical attributes, physical structures and chemical structures. It is possible to split the evaluation of durability into three objectives:

- **1.** To determine the mechanical degradation (e.g. toughness, tensile strength and modulus) of both un-aged and aged samples by tensile test.
- **2.** To evaluate chemical degradation in the matrix induced by thermooxidative aging by dynamic mechanical analysis, differential scanning calorimetry, and Fourier transform infrared spectroscopy tests.
- **3.** To verify physical degradation in the both un-aged and aged composites resulting from pronounced aging by weight loss and micrographic observations of topmost, cross sectional, and cryofractured surface.

1.4 Scopes of Work

The thesis will evaluate the influence that thermo-oxidative degradation has on the durability of epoxy composite that is reinforced by fiberglass from chemical, mechanical and physical points of view. The current work aims at linking the premature alterations that take place in the thin-skinned composite panel with reduction in durability. The examination of thermal-oxidative degradation through principles-based dimension taking into account the thickness of the original composite assists in the assessment of the real toughness behavior applicable in use-condition through mechanical testing. Consequently, assessment of toughness value through the calculation of the area that is below the stress-strain curve approach was employed over the Charpy impact test because of the sub-size thickness of the available materials. Additionally, the thesis introduces a methodology on the basis of industrial interest, as the thermo-oxidative aging of the thin-skinned composites remains under-researched. Thus, the study is expected to add to the existing literature on the materials employed in manufacturing aircrafts particularly in regard to their durability in relation to the marked thermaloxidative degradation.

1.5 Limitations

The main challenge in conducting the research likely to have limited the extent that FTIR and thermal analyses would cover relates to the failure by the manufacturers to reveal the compositional data and chemical structure of the matrix.

1.6 Research Contribution

The finding of this research work will redound to add to the existing literature on the EHG250-68-37 woven fiberglass/epoxy prepreg employed in manufacturing aircrafts particularly in regard to their thermal-oxidative degradation. In addition, the introduced methodology in this study will determine that which one of the physical aging and chemical aging has major role in the matrix embrittlement at the initial stage of thermo-oxidative aging (post curing period). This in turn, assists thin-skinned

polymer matrix composite manufacturers to take into account a proper curing schedule for PMCs laminates to avoid premature in-service failures.

1.7 Thesis Organization

In this thesis, Chapter 2 will provide the literature review covering the relevant published articles on the isothermal thermo-oxidative aging on the polymer matrix composites as well as the consequent physical, mechanical and chemical variations. Whereas, Chapter 3 presents the experimental methodology employed in evaluating the durability of the thin-sinned composites. On its part, Chapter 4 provides the results that the experimental tests and discussions obtained. Eventually, Chapter 5 provides the conclusions and recommendations that can be drawn from the undertaken research.



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