



**UNIVERSITI PUTRA MALAYSIA**

**IMPROVING BUCKLING AND POST-BUCKLING OF SHAPE MEMORY  
ALLOY LAMINATED COMPOSITE PLATES SUBJECTED TO MECHANICAL  
AND THERMAL LOADING USING FINITE ELEMENT METHOD**

**ZAINUDIN BIN A.RASID**

**FK 2012 81**

**IMPROVING BUCKLING AND POST-BUCKLING OF SHAPE MEMORY ALLOY  
LAMINATED COMPOSITE PLATES SUBJECTED TO MECHANICAL AND THERMAL  
LOADING USING FINITE ELEMENT METHOD**



**Thesis Submitted to the School of Graduate Studies, Universiti Putra Malaysia, in Fulfilment of  
the Requirements for the Degree of Doctor of Philosophy**

**November 2012**

Abstract of thesis presented to the Senate of Universiti Putra Malaysia in fulfilment of the requirement  
for the degree of Doctor of Philosophy

**IMPROVING BUCKLING AND POST-BUCKLING OF SHAPE MEMORY ALLOY  
LAMINATED COMPOSITE PLATES SUBJECTED TO MECHANICAL AND THERMAL  
LOADING USING FINITE ELEMENT METHOD**

By

**ZAINUDIN BIN A.RASID**

**November 2012**

**Chairman: Assoc. Prof. Rizal Zahari, PhD**

**Faculty: Engineering**

This research work focuses on simulation work involving development of finite element formulations and its finite element based software validated against experimental results reported in the literature to subsequently facilitate parametric studies. Nitinol shape memory alloy with its well-known property of the shape memory effect is used to improve post-buckling of laminated composite plates subjected to mechanical, thermal and thermo-mechanical loadings. Two finite element formulations for the post-buckling of composite plates with embedded shape memory alloy, namely the total strain and the incremental strain formulations are used. Both formulations are derived based on the first order shear deformation theory while the strength of material approach is used to include the effect of recovery stress in the constitutive equation. Thermal loading can be uniform or non-uniform throughout the width and thickness of the composite plates. The properties and recovery stress of the nitinol are either determined by solving the Brinson's model or taken from experimental data of others. The formulations were solved using the Newton-Raphson's method and source codes were developed for this purpose. Parametric studies were conducted theoretically to investigate the effects of the shape memory alloy on the post-buckling behaviour of composite plates with regard to several composite related and shape

memory alloy related parameters. The addition of shape memory alloy wires within layers of composite plates has resulted in the significant improvement in the composite critical loads. In the case of simply supported boundary condition, the increase of the critical load can be up to 70% for the shape memory alloy layer thickness equal to one fourth of the total thickness of other layers. The post-buckling paths of the composite plates subjected to mechanical, thermal and thermo-mechanical loadings are stable and substantially improved after the addition of the shape memory alloy. For the four types of configurations under studied here, the improvement of the active strain energy tuning method is at the highest in the case of the symmetric angle-ply plate where bifurcation for this plate occurs at the ratio of the load over critical load of  $P/P_{cr} = 3$ . It is interesting also to see that while the best mechanical post-buckling paths occur if the shape memory alloy layer is located in the middle of the plate, the location of the shape memory alloy layers has no effect on the thermal post-buckling paths. In the case of the tent-like temperature distribution, the non-uniform temperature distribution where the ratio of the temperature of the uniform temperature rise part to the temperature gradient,  $T_1/T_0 = 1$  has allowed the post-buckling response to occur earlier compared to the case of  $T_0 = 0$  loading. At the same time for both cases of the active property tuning and the active strain energy tuning, the post-buckling paths are improved with the increase of the ratio  $T_1/T_0$ . Furthermore, the thermal post-buckling paths that are degraded initially due to the compressive loading are shown to jump upward significantly with the addition of the shape memory alloy. At the end, this research has shown that the developed model and the source codes are able not only to show the significant improvement made by the shape memory alloy on the post-buckling behaviour of composite plates subjected to mechanical, thermal and thermo-mechanical loadings but also to demonstrate the post-buckling behaviour of the shape memory alloy composite plates subjected to several parameter changes.

Abstrak tesis yang dikemukakan kepada Senat Universiti Putra Malaysia sebagai memenuhi keperluan untuk ijazah Doktor Falsafah

**PENAMBAHBAIKAN KEPADA LENGKOKAN AND PASCALENGKOKAN BAGI PLAT KOMPOSIT TERLAMINAT DENGAN ALOI MEMORI BENTUK YANG DIKENAKAN BEBANAN MEKANIK DAN HABA MENGGUNAKAN KAEDAH UNSUR TERHINGGA**

Oleh

**ZAINUDIN BIN A.RASID**

**November 2012**

**Pengerusi: Prof. Madya Rizal Zahari, PhD**

**Fakulti: Kejuruteraan**

Dalam kajian ini, aloi memori bentuk nitinol dengan kelakuan kesan memori bentuknya yang terkenal itu telah digunakan untuk memperbaiki kelakuan pascalengkokan bagi plat komposit berlapis yang dikenakan bebanan mekanik, haba dan haba mekanik. Dua perumusan unsur terhingga untuk kelakuan pascalengkokan bagi plat komposit dengan aloi memori bentuk terbenam iaitu formulasi terikan jumlah dan formulasi terikan tokokan telah digunakan. Kedua-dua model telah diterbitkan berdasarkan teori ubah bentuk rincih tertib pertama di mana pendekatan kekuatan bahan telah digunakan bagi memasukkan kesan tegasan pemulihan ke dalam persamaan juzuk bagi plat komposit. Bebanan haba adalah termasuklah yang seragam atau tidak seragam di sepanjang keluasan dan ketebalan plat. Kelakuan dan tegasan pemulihan bagi aloi memori bentuk ditentukan dengan menyelesaikan model Brinson ataupun dengan menggunakan data-data ujikaji pengkaji lain. Kedua-dua model telah diselesaikan menggunakan kaedah Newton-Raphson dan kod sumber telah dibangunkan untuk tujuan ini. Kajian berparameter telah dilakukan secara teori dalam menyiasat kesan aloi memori bentuk ke atas kelakuan pascalengkokan bagi plat komposit berlapis merujuk kepada beberapa kelakuan yang berkaitan dengan komposit dan kelakuan yang berkaitan dengan aloi memori bentuk. Penambahan aloi

memori bentuk di antara lapisan plat komposit telah menghasilkan penambahbaikan yang bererti kepada beban kritikal. Untuk kes keadaan sempadan plat disokong mudah, penambahan beban kritikal sebanyak 70% terhasil bila ketebalan lapisan aloi memori bentuk adalah satu perempat ketebalan lapisan yang berbaki. Laluan pascalengkokan bagi plat komposit adalah didapati stabil dengan penambahbaikan yang bererti bagi semua kes-kes bebanan mekanik, haba dan mekanik haba setelah aloi memori bentuk dibenam. Bagi empat jenis tatarajah yang dikaji di sini, penambahbaikan adalah tertinggi dalam kes plat lapis-serong simetri di mana dwi-kewujudan berlaku pada nisbah beban kepada beban kritikal,  $P/P_{cr} = 3$ . Adalah menarik untuk menyaksikan laluan pascalengkokan bagi kes bebanan haba tidak dipengaruhi oleh kedudukan lapisan aloi memori bentuk sedangkan bagi kes bebanan mekanik, laluan yang terbaik adalah bila lapisan aloi memori bentuk berada di tengah-tengah plat. Bagi kes bebanan taburan suhu seperti khemah, taburan suhu yang tak seragam di mana nisbah suhu bahagian tak seragam kepada suhu bahagian tetap,  $T_1/T_0 = 1$  telah membenarkan tindakbalas pascalengkokan berlaku lebih awal berbanding kes  $T_0 = 0$ . Pada masa yang sama, bagi kedua-dua kes penalaan sifat aktif dan penalaan tenaga terikan aktif, laluan pascalengkokan bertambah baik dengan penambahan nisbah suhu  $T_1/T_0$ . Tambahan pula, lalaun pascalengkokan yang telah terkurang akibat bebanan mekanik yang telah dikenakan lebih awal telah didapati mengalami anjakan yang besar ke arah kedudukan lebih baik dengan penambahan aloi memori bentuk. Akhir sekali, kajian ini telah menunjukkan bahwa model-model yang telah dibangunkan bersama kod sumbernya telah bukan sahaja menzahirkan penambahbaikan yang bererti yang dibuat oleh aloi memori bentuk terhadap kelakuan pascalengkokan plat komposit berlapis yang dikenakan beban mekanik, haba dan haba-mekanik bahkan juga memperlihatkan kelakuan pascalengkokan plat terhadap beberapa perubahan parameter.

## **ACKNOWLEDGEMENTS**

I would like to express my sincere gratitude and deep thanks to my supervisor Assoc. Prof. Dr. Rizal Zahari for his kind assistance, support, advice, encouragement and suggestions throughout this research and during the preparation of this thesis.

A particular note of thanks is also given to members of supervisory committee, Dr. Dayang Laila Abang Abdul Majid and Dr. Azmin Shakrine Mohd Rafie for their suggestions and constructive criticisms given at stages of this study.

My ardent pray is intended to my late father, Hj Abdul Rashid Md Yasin and my late mother, Hjh. Zainun Hj Ali for their past love and prays that have pushed me forward during difficult time.

And great thanks to my wife, Zukarni Hashim and my six children: Ihsan, Nasuha, Iman, Aisyah, Umar and Sarah for their patient and perseverance.

I certify that a Thesis Examination Committee has met on **the 19<sup>th</sup> November 2012** to conduct the final examination of Zainudin bin A. Rasid on his thesis entitled "**Improving Buckling and Post-Buckling of Shape Memory Alloy Laminated Composite Plates Subjected to Mechanical and Thermal Loading Using Finite Element Method**" in accordance with the Universities and University Colleges Act 1971 and the Constitution of the Universiti Putra Malaysia [P.U. (A) 106] 15 March 1998. The Committee recommends that the student be awarded the Doctor of Philosophy.

Members of the Thesis Examination Committee were as follows:

**Renuganth A/l Varatharajoo, PhD, P. Eng.**

Professor

Faculty of Engineering  
Universiti Putra Malaysia  
(Chairman)

**Mohd Sapuan Salit, PhD, P. Eng.**

Professor

Faculty of Engineering  
Universiti Putra Malaysia  
(Internal Examiner)

**Faizal Bin Mustapha, PhD**

Associate Professor

Faculty of Engineering  
Universiti Putra Malaysia  
(Internal Examiner)

**Mohammed Sarwar Janghashmi, PhD**

Professor

Faculty of Mechanical and Manufacturing  
Dublin City University  
Ireland  
(External Examiner)

---

**PROF. DR. SEOW HENG FONG**

Professor and Deputy Dean  
School of Graduate Studies  
Universiti Putra Malaysia

Date: 19<sup>th</sup> December, 2012

This thesis was submitted to the senate of Universiti Putra Malaysia and has been accepted as fulfilment of the requirement for the degree of Doctor of Philosophy. The members of the Supervisory Committee were as follows:

**Rizal Zahari, PhD**

Associate Professor

Faculty of Engineering

Universiti Putra Malaysia

(Chairman)

**Azmin Shahrain Mohd Rafie, PhD**

Senior lecturer

Faculty of Engineering

Universiti Putra Malaysia

(Member)

**Dayang Laila Abang Abdul Majid, PhD**

Senior lecturer

Faculty of Engineering

Universiti Putra Malaysia

(Member)

---

**BUJANG BIN KIM HUAT, PhD**

Professor and Dean

School of Graduate Studies

Universiti Putra Malaysia

Date:

## **DECLARATION**

I declare that the thesis is my original work except for quotations and citations which have been duly acknowledged. I also declare that it has not been previously, and is not concurrently, submitted for any other degree at Universiti Putra Malaysia or at any other institution.



**ZAINUDIN A. RASID**

Date: 19<sup>th</sup> November, 2012

## TABLE OF CONTENTS

	Page
<b>ABSTRACT</b>	ii
<b>ABSTRAK</b>	iv
<b>ACKNOWLEDGEMENTS</b>	vi
<b>APPROVAL</b>	vii
<b>DECLARATION</b>	ix
<b>LIST OF TABLES</b>	x
<b>LIST OF FIGURES</b>	xi
<b>LIST OF ABBREVIATIONS</b>	xvii
 <b>CHAPTER</b>	
1 <b>INTRODUCTION</b>	1
1.1 Background	1
1.2 Problem statements	10
1.3 Hypothesis	11
1.4 Objectives	11
1.5 Scope	11
1.6 Applications	13
1.7 Thesis organization	15
2 <b>LITERATURE REVIEW</b>	18
2.1 Laminated Composite	18
2.2 Thermo-mechanical Post-buckling of Composite Plates	20
2.2.1 Mechanical loading	20
2.2.2 Thermal loading	23
2.2.3 Thermo-mechanical loading	25
2.3 The effects of SMA on structural behaviours	26
2.3.1 Behaviours of the embedded SMA	27
2.3.2 Control strategies and SMA configuration	29
2.3.3 Predicting the SMA behaviours	32
2.3.3.1 SMA constitutive models	32
2.3.3.2 Experimental prediction of Recovery Stress	36
2.4 Post-buckling improvement of composite plates using SMA	38
2.5 Summary	45

<b>3</b>	<b>MATERIALS AND METHODS</b>	47
3.1	Material	47
3.1.1	Material properties	47
3.1.2	Configuration of the SMA composite plates	49
3.1.3	Recovery stress	53
3.1.3.1	Experimental recovery stress	53
3.1.3.2	Recovery stress from the Brinson's model	55
3.1.3.3	Material parameter	57
3.1.3.4	Results from the Brinson's model	57
3.2	Methods	60
3.2.1	Total strain formulation	60
3.2.1.1	Effective properties	61
3.2.1.2	Stress-strain relationship	66
3.2.1.3	Displacement field and strain	68
3.2.1.4	Stress resultants	69
3.2.1.5	Finite element implementation	72
3.2.1.6	Principle of virtual work	74
3.2.2	Incremental strain formulation	78
3.2.2.1	Displacement field and strain	79
3.2.2.2	Stress resultant	80
3.2.2.3	Finite element implementation	82
3.2.2.4	Principle of virtual work	84
3.3	Source code development	89
3.4	Summary	94
<b>4</b>	<b>RESULTS AND DISCUSSIONS</b>	95
4.1	Validation	95
4.1.1	Linear buckling analyses of SMA composite plates	95
4.1.2	Mechanical post-buckling analysis	97
4.1.3	Thermal post-buckling analysis	97
4.1.4	Thermal post-buckling analysis due to non-uniform temperature distribution	99
4.2	The Linear Buckling Analysis	100
4.2.1	The effect of the thickness of the SMA layer	100

4.2.2 The effect of the composite number of layers	103
4.2.3 The effect of the angle of orientations	105
4.2.4 The effect of the composite configurations	106
4.2.5 The effect of the location of the SMA layers	107
4.2.6 The effect of the volume fraction of nitinol	109
4.2.7 The effect of the activation temperature of the SMA wires	110
4.2.8 The effect of the initial strain of the SMA wires	112
4.2.9 Summary of the linear buckling	113
4.3 The Mechanical Post-buckling Analysis	115
4.3.1 The effect of the composite configurations	115
4.3.2 The effect of the location of the SMA layers on critical loads	120
4.3.3 The effect of the SMA activation temperatures	123
4.3.4 The effect of the SMA volume fractions	125
4.3.5 The effect of the SMA initial strain	126
4.3.6 Summary of the mechanical post-buckling analysis	128
4.4 The Thermal Post-buckling Analysis	129
4.4.1 The effect of the boundary conditions	129
4.4.2 The effect of the SMA volume fraction	131
4.4.3 The effect of the initial strain	132
4.4.4 The effect of the composite configurations	133
4.4.5 The effect of the number of layers of the composite plate	135
4.4.6 The effect of the location of the SMA layers	136
4.4.7 Thermal post-buckling due to tent-like temperature distribution	137
4.4.8 Thermal post-buckling due to parabolic and linearly varied through thickness temperature distributions	142
4.4.9 Summary of the thermal post-buckling analysis	148
4.5 The Thermo-mechanical Post-buckling Analysis	150
4.5.1 The thermo-mechanical post-buckling of composite plates without SMA	151
4.5.2 The SMA post-buckling improvement	153
4.5.3 Summary of the thermo-mechanical post-buckling analysis	157
4.6 Summary	157

<b>5. SUMMARY, CONCLUSION AND RECOMMENDATIONS FOR FUTURE RESEARCH</b>	<b>159</b>
5.1 Summary	159
5.2 Conclusion	161
5.3 Recommendations	161
<b>REFERENCES</b>	<b>164</b>
<b>BIODATA OF STUDENT</b>	<b>171</b>
<b>LIST OF PUBLICATIONS</b>	<b>172</b>

